

TO: ALL HOLDERS OF THRUST REVERSER TEST PANEL ASSEMBLY M1977 COMPONENT MAINTENANCE MANUAL 78-35-02

REVISION NO. 2 DATED JAN 01/93

HIGHLIGHTS

Pages which have been added or revised are outlined below together with the highlights of the revision. Remove and insert the affected pages as listed and enter Revision No. and date on the Record of Revision Sheet.

CHAPTER/SECTION

AND PAGE NO. DESCRIPTION OF CHANGE

101 Added a current limit specification to the power

supply.

102 Added a component location configuration.

103-105 Changed the switch nomenclatures shown in the

functional test procedure and the schematic diagram.



THRUST REVERSER TEST PANEL ASSEMBLY, M1977

PART NUMBER 233T4259-1

COMPONENT MAINTENANCE MANUAL WITH ILLUSTRATED PARTS LIST



REVISION RECORD

• Retain this record in front of manual. On receipt of revision, insert revised pages in the manual, and enter revision number, date inserted and initial.

REVISION NUMBER	REVISION DATE	DATE FILED	ВҮ	REVISION NUMBER	REVISION DATE	DATE FILED	вү

REVISION RECORD
Page 1



TEMPORARY REVISION AND SERVICE BULLETIN RECORD

BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVE	DATE OF INCORPORATION INTO MANUAL
		PRR B11480-107	JUL 01/92



PAGE	DATE	CODE	PAGE	DATE	CODE
			1	PARTS LIST	
78-35-02			1001	JUL 01/92	01
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*104	JAN 01/93	01.1			
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^{* =} REVISED, ADDED OR DELETED

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Repair	601
Assembly	
Fits and Clearances (not applicable)	
Special Tools	
Illustrated Parts List	1001
*[1] Use applicable procedures in 20—11—04 and standard industry practices *[2] Special instructions not required.	i -



INTRODUCTION

The instructions in this manual provide the information necessary to perform maintenance functions including test, fault isolation, and replacement of defective components.

This manual is divided into separate sections:

- 1. Title Page
- 2. Record of Revisions
- 3. Temporary Revision & Service Bulletin Record
- 4. List of Effective Pages
- 5. Table of Contents
- 6. Introduction
- 7. Procedures & IPL Sections

Refer to the Table of Contents for the page location of applicable sections.

An explanation of the use of the Illustrated Parts List is provided in the Introduction to that section.

All weights and measurements used in the manual are in English units, unless otherwise stated. When metric equivalents are given they will be in parentheses following the English units.

Design changes, optional parts, configuration differences and Service Bulletin modifications create alternate part numbers. These are identified in the Illustrated Parts List (IPL) by adding an alphabetical character to the basic item number. The resulting item number is called an alpha-variant. Throughout the manual, IPL basic item number references also apply to alpha-variants unless otherwise indicated.

Jul 01/92



THRUST REVERSER TEST PANEL ASSEMBLY M1977

DESCRIPTION AND OPERATION

1. <u>Description</u>

A. The thrust reverser test panel assembly contains indicator lights and switches mounted on a baseplate. External connections are made through four connectors mounted on a support bracket in rear.



TESTING AND TROUBLE SHOOTING

1. <u>Test Equipments</u>

A. Power Supply: 28 ±1v dc, 0.5A max

B. Multimeter: Simpson 260 or equivalent

C. Test Connector: Boeing breakout box A33003-2 with test cables shown below may be used

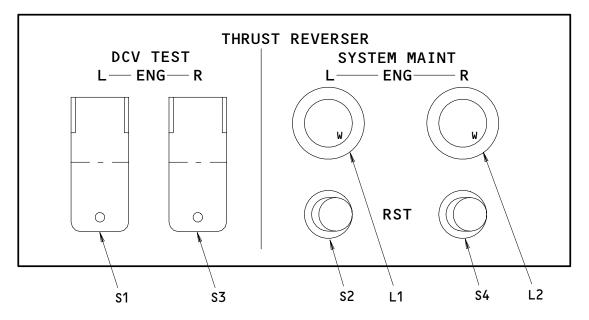
<u>Test Connector</u>	<u>Mates with</u>	<u>Test Cable</u>
BACC45FT8-3S	J1	_
BACC45FT10-5S	J2	A33003-160
BACC45FT14-15S	J3	A33003-18
BACC45FT12-12S	J4	A33003-9

2. Functional Test

- A. Connect the thrust reverser test panel assembly to the test connectors.
- B. Do the functional test procedure shown in Fig. 101.

NOTE: Continuity is defined as 3 ohms or less and Open is defined as 900k ohms or greater.





233T4259-1 Component Location Figure 101



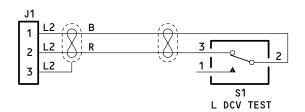
	TEST STEP	PROCEDURE	REQUIRED RESUL	т	COMPONENT TEST
1	1	Measure ohms between J1-1 and J1-2 with S1 (L DCV TEST) set to the position specified in the required result column.		Released Continuity	S1
1	2	Measure ohms between J2-5 and J2-4 with S3 (R DCV TEST) set to the position specified in the required result column.	,	Released Continuity	S 3
I	3	Apply 28v dc to J3-5 and ground to J3-10.	L1 (L SYSTEM MAINT)	on	L1
	4	Momentarily press and release L1 (L SYSTEM MAINT).	L1 off while pressed	I	L1
	5	Remove ground from J3-10.			
I	6	Ground J3-7.	L1 (L SYSTEM MAINT)	off	L1
	7	Momentarily press and release L1 (L SYSTEM MAINT).	L1 on while pressed		L1
	8	Remove 28v dc from J3-5 and ground from J3-7.			
	9	Measure ohms between J3-15 and J3-12 with S2 (L RESET) set to the position specified in the required result column.		<u>Released</u> Open	s 2

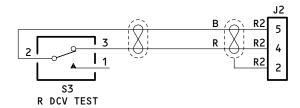
Functional Test Procedure Figure 102 (Sheet 1)

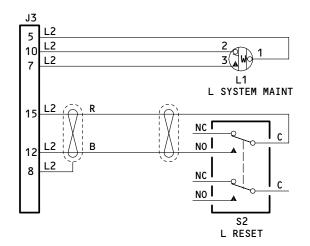


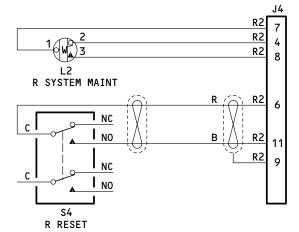
	TEST STEP	PROCEDURE	REQUIRED RESULT	COMPONENT TEST
I	10	Apply 28v dc to J4-7 and ground to J4-4.	L2 (R SYSTEM MAINT) on	L2
1	11	Momentarily press and release L2 (R SYSTEM MAINT).	L2 off while pressed	L2
	12	Remove ground from J4-4.		
ı	13	Ground J4-8.	L2 (R SYSTEM MAINT) off	L2
1	14	Momentarily press and release L2 (R SYSTEM MAINT).	L2 on while pressed	L2
	15	Remove 28v dc from J4-7 and ground from J4-8.		
	16	Measure ohms between J4-6 and J4-11 with S4 (R RESET) set to the position specified in the required result column.	<u>Pressed & Held</u> <u>Released</u> Continuity Open	\$4
	17	Remove all connections.		

Functional Test Procedure Figure 102 (Sheet 2)









NOTE: ALL WIRES BMS 13-48 TYPE 1, CLASS 1, AND TYPE 24, CLASS 2, AWG 20 EXCEPT AS NOTED. SEE REPAIR FOR WIRE SEPARATION.

Schematic Diagram Figure 103



REPAIR

- 1. All repair may be accomplished with standard industry practices and procedures contained in 20-11-04 except as follows:
 - A. Wire categories are color-coded R2 (green) and L2 (red). Maintain a minimum of 1/4-inch spacing between bundles of different categories.

REPAIR-GENERAL



ILLUSTRATED PARTS LIST

- This section lists and illustrates replaceable or repairable component parts. The Illustrated Parts Catalog contains a complete explanation of the Boeing part numbering system.
- 2. Indentures show parts relationships as follows:

Assembly
Detail Parts for Assembly
Subassembly
Attaching Parts for Subassembly
Detail Parts for Subassembly

Detail Installation Parts (Included only if installation parts may be returned to shop as part of assembly)

- 3. One use code letter (A, B, C, etc.) is assigned in the EFF CODE column for each variation of top assembly. All listed parts are used on all top assemblies except when limitations are shown by use code letter opposite individual part entries.
- 4. Letter suffixes (alpha-variants) are added to item numbers for optional parts, Service Bulletin modification parts, configuration differences (Except left- and right-hand parts), product improvement parts, and parts added between two sequential item numbers. The alpha-variant is not shown on illustrations when appearance and location of all variants of the part is the same.
- 5. Service Bulletin modifications are shown by the notations PRE SB XXXX and POST SB XXXX.
 - A. When a new top assembly part number is assigned by Service Bulletin, the notations appear at the top assembly level only. The configuration differences at detail part level are then shown by use code letter.
 - B. When the top assembly part number is not changed by the Service Bulletin, the notations appear at the detail part level.

6. Parts Interchangeability

Optional The parts are optional to and interchangeable (OPT) with other parts having the same item number.

Supersedes, Superseded By The part supersedes and is not interchangeable (SUPSDS, SUPSD BY) with the original part.

Replaces, Replaced By

The part replaces and is interchangeable with, (REPLS, REPLD BY)

or is an alternate to, the original part.



7. Use substitute parts only as specified in the drawing, 280T1001, Airplane Electrical/Electronic Parts Substitution Drawing.



VENDORS

DEXTER-WILSON CORP
4133 STONE WAY NORTH
SEATTLE, WASHINGTON 98103-8013

81590 KORRY ELECTRONIC INC SUB OF CRITON CORP
901 DEXTER AVENUE NORTH
SEATTLE, WASHINGTON 98109-3515

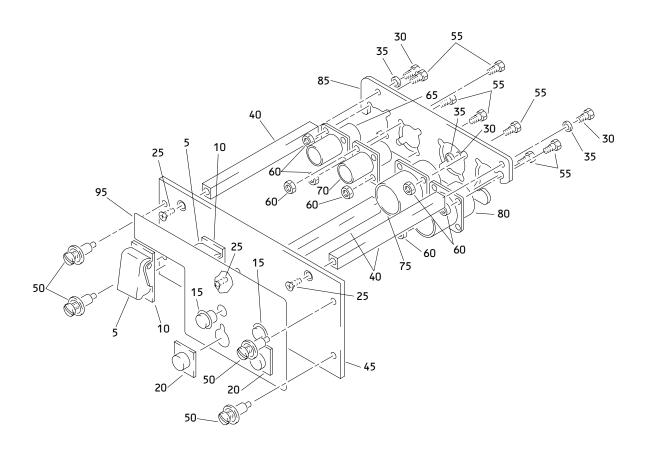
91929 HONEYWELL INC MICRO SWITCH DIV
11 WEST SPRING STREET

FREEPORT, ILLINOIS 61032



REFERENCE DESIGNATOR INDEX (SEE SCHEMATIC DIAGRAM)				
REFERENCE DESIGNATOR	PART NUMBER	FIG-ITEM		
J1	BACC45FN8-3P	1-65		
J2	BACC45FN10-5P	70		
J3	BACC45FN14-15P	75		
J4	BACC45FN12-12P	80		
L1,L2	BCREFA3292	15		
S1	MS24523-26	10		
S 2	2PB11H58	20		
S 3	MS24523-26	10		
\$4	2РВ11Н58	20		





Thrust Reverser Test Panel Assembly, M1977 Figure 1

78–35–02

& ITEM	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1234567	EFF CODE	QTY PER ASSY
01- -1	233T4259-1		PANEL ASSY-THRUST REVERSER		RF
i '	23314237 1		TEST, M1977		IXI
5	400-122		.GUARD-SWITCH		2
			(V28277)	1	
10	MS24523-26		.SWITCH		2
			(S1, S3)		
15	BACREFA3292		.LIGHT-IND		2
			(V81590)		
			(116–565–1042–002)		
20	2PB11H58		(L1, L2)		2
20	2761100		.SWITCH-PUSHBUTTON (V91929)		۷
			(\$2, \$4)		
25	NAS514P632-5P		.SCREW		3
1 1	NAS1801-06-5		SCREW		3
1 1	MS35338-41		.WASHER		3 3 3 3
1 1	69B46200-7		.STANDOFF		3
45	233T4259-2		.BASEPLATE ASSY		1
50	BACS21DX1B		STUD ASSY		4
55	NAS1801-04-5		.SCREW		8
	BACN1ONW1		. NUT		8
65	BACC45FN8-3P		_ CONNECTOR		1
			(J1)		_
70	BACC45FN10-5P		- CONNECTOR		1
	D100/FFN4/ 455		(J2)		
75	BACC45FN14-15P		- CONNECTOR		1
80	BACC45FN12-12P		(J3) .CONNECTOR		1
00	DACC4JFN12-12P		(J4)		'
85	233T6200-197		.SUPPORT-CONNECTOR		1
1 1	233T4259-3		.WIRE BUNDLE ASSY		1
1 - 1	BAC27TFDE0438		.DECAL		1